

Modelling and simulation of flame cutting for steel plates with solid phases and melting

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Abstract

The goal of this work is to describe in detail a quasi-stationary state model which can be used to deeply understand the distribution of the heat in a steel plate and the changes in the solid phases of the steel and into liquid phase during the flame cutting process. We use a 3D-model similar to previous works from Thiébaud [1] and expand it to consider phases changes, in particular, austenite formation and melting of material. Experimental data is used to validate the model and study its capabilities. Parameters defining the shape of the volumetric heat source and the power density are calibrated to achieve good agreement with temperature measurements. Similarities and differences with other models from literature are discussed.

1 Introduction

In industry, the flame cutting process occurs in the final stage of the processing of steel plates. By then, the desired properties of the steel have already been established with different heat treatment processes. Hence the goal is to perform the cutting process without altering the mechanical and chemical properties already settled. In the case of thermal cutting, a thin section is removed, named kerf, and a heat affected zone (HAZ) is created around the cutting edge of the remaining pieces, which can cause deviations to the expected properties of the steel owing to occurring solid-solid phase transitions in the HAZ, which will result in changes in the thermomechanical properties. Moreover, for certain process parameters it has been found that cold cracks can appear a couple of days after the cutting possibly due to tensile stress, hydrogen content and microstructural effects [2].

A deeper knowledge of the cutting process can help to identify the crucial factors to maintain the good quality of the steel plates. This can be done in a suitable and non-expensive way with the help of computer simulations after the physics of the problem has been understood. In contrast to welding, which is a similar physical process, there are very few publications studying thermal cutting from a modelling and numerical simulation point of view.

In Lindgren et al. [3] the authors use a 2D-model to study the temperature evolution in the section transverse to the direction of the flame cutting. The heat input is adjusted with experimental measurements of two isothermal lines. The paper reveals the decrease in residual stress with a pre-heating stage of the plate prior to cutting using a model for mechanical analysis. Jokiahio et al. [4] employ the commercial finite element software ABAQUS to model the flame cutting of thick plates. Again, a 2D-model is used to register the nodal temperature history in the transverse plane. Different values for the heat input are used for different cutting speeds. The aforementioned values are calibrated such that the temperatures in the middle plane of the plate (half the thickness) do not reach the melting point. The study focuses on the stress analysis under different operating conditions: thickness, speed,

pre-heating stage. Bae et al. [5] work with ABAQUS for a 2D-model. The authors adopt a more complex definition of the heat flux. It is divided in three parts: upper surface due to the reaction between ethylene and oxygen, cutting line due to the reaction between oxygen and iron, and bottom of cutting line due to the heat conducted by the molten droplets. The model predicts the shape of the HAZ for diverse plate thicknesses, cutting speeds and flow rates of oxygen/ethylene. Thiébaud et al. [1] study the quasi-steady state of the temperature distribution during flame cutting with a 3D-model. The heat input is modelled as a volumetric heat source with a cylindrical-like shape. The simulation domain contains a gap related to an assumed kerf width. Two parameters are used to match the temperature profiles measured by several thermocouples on the top and bottom of the plate. These parameters are the heat density within the flame and the heat transfer coefficient in the aforementioned gap. Good matching with the experimental data is achieved.

In [6], a sophisticated multi-physics 3D-model is considered including gas dynamics and the melting of steel. However, the authors conclude that the computational effort for such a complex model is too costly for current hardware.

The present study is based on a quasi-stationary state 3D-model similar to the one used by Thiébaud. Using experimental data, the heat input power of the flame is adjusted to fit the data in the best possible way. The main novelty of the paper is the introduction of a quasi-stationary state (QSS) model comprising an energy balance coupled with two transport equations to model the occurring solid-liquid and solid-solid phase transitions. This approach allows for a more precise estimation of kerf and the heat affected zone.

The document is organized as follows. In Section 2, we describe a mathematical model for flame cutting of steel plates, consisting of a quasi-stationary heat equation coupled with a set of equations describing the phase transitions in steel. The considerations for numerical approximation and material data are presented in Section 3. Section 4 is devoted to the presentation of the results and comparison with experimental data used for validation followed by some concluding remarks in Section 5.

2 Modelling

2.1 Domain

Flame cutting of steel plates consists of removing a small amount of material from the plate along a straight cutting line dividing it in two pieces and leaving them with a very regular edge. For symmetry reasons, only one half of the plate will be considered, denoted by $\widetilde{\Omega}$. Due to the big size of the steel plate in comparison to the flame and the straight cutting trajectory, later on we will describe the cutting process by a quasi-stationary state (QSS) model. The plate moves with velocity v relative to the stationary torch. We define a cut-out Ω around the torch, which is represented by the subdomain Λ in Figure 1. The symmetry plane is $\{y = 0\}$.

2.2 Heat Source

The burning and removal of material is a complex process that can be described in three steps:

- i) A torch located on top of the plate raises the steel to its ignition temperature ($\sim 816^\circ\text{C}$). For this, the torch creates a flame using a mixture of oxygen and a fuel gas, commonly acetylene, butane or propane.

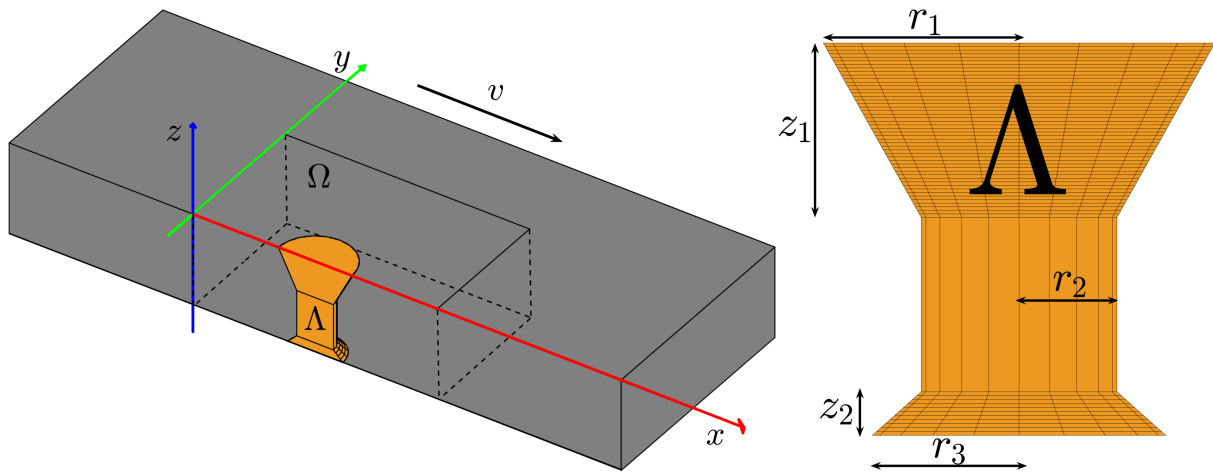


Figure 1: Half of steel plate and flame representation.

- ii) The iron in the steel reacts with the highly concentrated oxygen from the torch in an exothermic oxidation reaction adding more heat to the plate.
- iii) The steel below the torch reaches melting temperature ($\sim 1537^{\circ}\text{C}$). By heat conduction the melting temperature reaches the bottom of the plate. Immediately, the molten steel is flushed away by the high pressure oxygen stream.

These steps occur within a matter of seconds and additionally, the plate is moving in a fixed direction, thus separating the workpiece in two. The resultant edges of each new half of the steel are called cutting edges and the gap created is named kerf or groove. The process is ruled by two main factors: the cutting speed v and the flame. Depending on the steel properties and the thickness of the plate, a suitable speed must be chosen by the operators in the factory. The shape of the flame is determined by the torch used while the power depends on the combustion gas selected and the pressure of the oxygen and fuel gas stream.

Mathematically, the heat source $Q(v)$ can be characterized in terms of the shape of the flame created by the torch. It is defined as a constant power density within a particular volume Λ and assumed to be zero outside of it. More precisely, the power density is determined by a constant value for power P divided by the volume of Λ . The heat source is uniformly distributed within the volume Λ , which is similar to a cylinder whose radius changes with height, specifically Λ can be identified as a cylinder with a truncated cone on the top and at the bottom (see Figure 1). The variation of shape, and therefore heat, depending on height is based on physical reasons. While on the cutting plane most of the heat comes from the oxidation of iron and concentrates in a thin cylindrical volume below the torch, extra heat input exists close to the top and bottom surfaces of the plate: at the bottom by the presence of droplets of molten metal and on the top surface by the reaction between oxygen and fuel gas [1, 3, 4, 5].

In Figure 2 two sets of isothermal curves in cross sections parallel and with growing distance to the cutting surface are depicted. The sets correspond to a speed of 135 and 270 mm/min, respectively, during the flame cutting process. The isothermal lines were estimated using the temperature records from thermocouples (attached to holes drilled in the plate), microstructure study and literature. Both sets start in the cutting edge of the remaining half after flame cutting. The cutting edge is almost aligned with the 1537°C curves. Its shape can be regarded as an experimental evidence for the assumed shape of the volume Λ . Moreover, higher temperatures are reached further within the plate in the slower case as the heat source has more influence and due to more time for heat conduction.

This fact can also be seen in Jokiaho et al. [7], where a conclusion of the flame cutting experiments is that the heat affected zone width depends on cutting speed.

Furthermore, it is important to mention that the kerf (removed area) in both cases is different, being smaller in the faster case. In Duan et al. [8], it is stated that the kerf width is generally assumed to be constant in most mathematical models of flame cutting. However, a large error may be produced if this variation in width with respect to thickness and cutting speed is neglected. Hence, we assume that the kerf size decreases with increasing speed and model this effect by changing the radius of Λ as a function of velocity v .

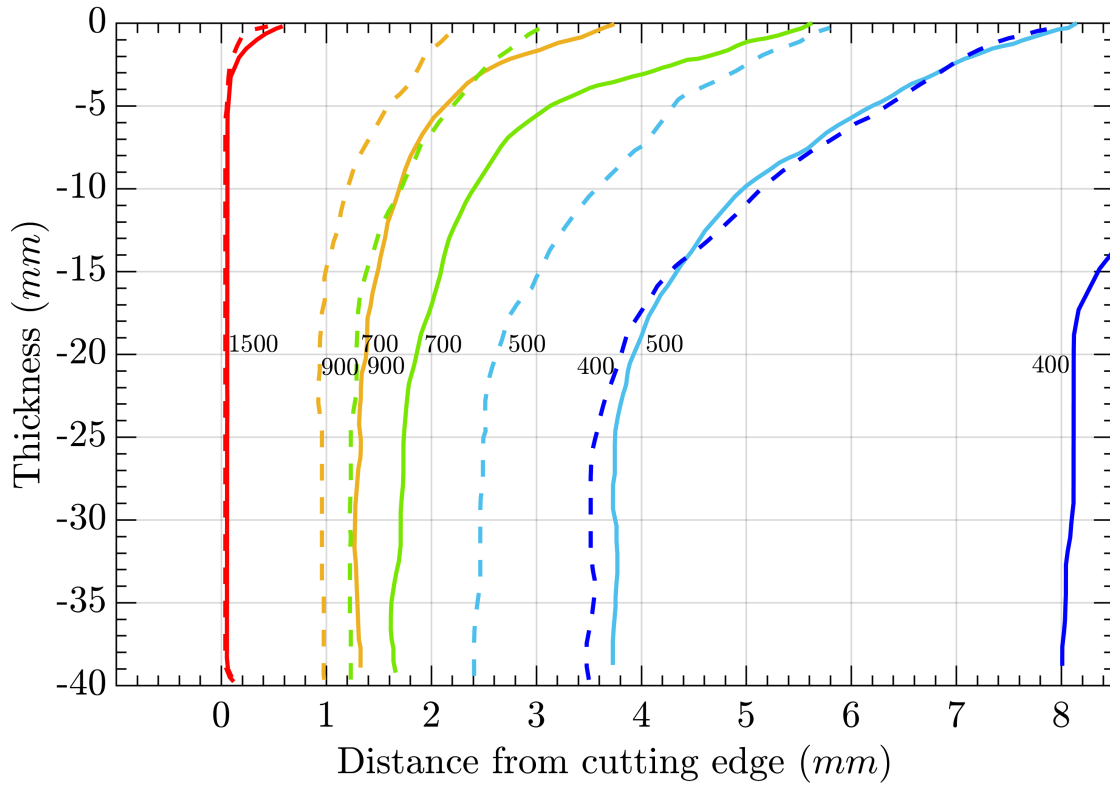


Figure 2: Isothermal lines in the plane transversal to the cutting direction. Case 135 mm/min (solid) and case 270 mm/min (dashed).

In view of all these considerations, we define Λ as

$$\Lambda(\mathbf{v}) = \left\{ \mathbf{x} \in \tilde{\Omega} \mid (x - x_{\Lambda})^2 + (y - y_{\Lambda})^2 \leq R(z, \mathbf{v})^2, -h \leq z \leq 0 \right\}. \quad (1)$$

Here, $(x_{\Lambda}, y_{\Lambda})$ are the coordinates of the symmetry axis of Λ . The simplest way to model a source with the above mentioned phenomena is with only five parameters that define the shape of the heat source along the complete thickness h of the plate: r_1 , r_2 , and r_3 are the upper, medium and lower radii, respectively, and z_1 and z_2 the heights of the top and bottom truncated cones, respectively (see Figure 1).

We define the function $R(z, \mathbf{v})$ from Equation 1 reflecting the radial change of Λ according to height and speed in a separable way as a product of two functions, i.e.,

$$R(z, \mathbf{v}) = C(\mathbf{v})\bar{R}(z), \quad (2)$$

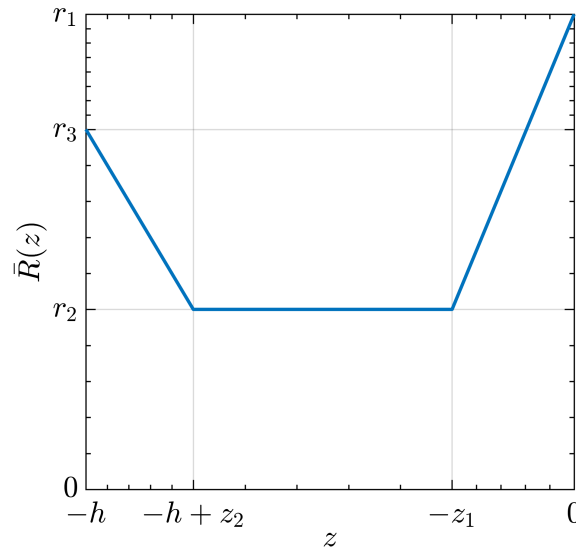


Figure 3: Definition of the function $\bar{R}(z)$ for the heat source.

$$\text{with } \bar{R}(z) = \begin{cases} \frac{r_2 - r_1}{\bar{z}_1} z + r_1, & z > -\bar{z}_1 \\ r_2, & -\bar{z}_1 \geq z \geq -h + \bar{z}_2 \\ \frac{r_2 - r_3}{\bar{z}_2} (z + h - \bar{z}_2) + r_2, & z < -h + \bar{z}_2 \end{cases} \quad (3)$$

The definition of $\bar{R}(z)$ manifest the difference in radii with height and can be seen in Figure 3. On the other hand, $C(v)$ takes into account that the faster the plate moves the smaller the area of influence of the flame gets. Thus, for convenience, $C(v)$ is assumed to be a linear, monotonically decreasing function that causes the radii of Λ to be scaled with velocity.

All in all, the heat source $Q(v)$ is then defined as

$$Q(v) = \frac{P}{|\Lambda(v)|} \chi_{\Lambda(v)} \quad (4)$$

with $\chi_{\Lambda(v)}$ the characteristic function of the volume $\Lambda(v)$.

2.3 Phase Transitions

During a heating process, steel undergoes microstructural changes affecting its physical properties. At room temperature, combinations of four different phases can be distinguished in the steel: ferrite, perlite, bainite and martensite. The presence and the proportion of each phase depends on the type of steel and its chemical composition, and on the heat treatment that the material has been subjected to. If the steel temperature grows sufficiently high, the previous solid phases are transformed into a new high temperature phase called austenite. This process is called austenization. During a cooling cycle, this austenite is dissolved into the previous phases. The volume fraction of the remaining phases is determined by the local cooling rate. For more detailed information we refer to [9].

During flame cutting, around the cutting plane, a portion of steel is molten and removed while in the neighbourhood of this kerf a Heat Affected Zone (HAZ) is formed. This HAZ is generated at the cutting edge of the steel plate due to large thermal gradients [7]. Phase transitions occur in the HAZ as the

austenization temperature is reached. The HAZ is an important region to study as it is the area where cracks have been found to appear, eventually leading to quality losses and customer complaints.

For modelling the flame cutting process it is important to account for the melting of steel and thus to estimate the size of the kerf. To this end, we add the relative fraction of molten steel as an extra phase to our phase transition model. In Hömberg et al. [10], a transient model based on the Leblond-Devaux model is proposed that reproduces the relative volume fractions of the different solid phases of steel during a heating and cooling cycle.

For the purpose of predicting the HAZ size, which will provide information about the possible occurrence of cracks, it is sufficient to keep track of the formation of austenite. With this in consideration, we propose a reduced model, based on [10], consisting of a rate law for austenite a and liquid phase (molten steel) l with θ being the temperature. The corresponding system of equations for these two phases is the following:

$$\dot{l} = \frac{1}{\tau_l} [l_{eq}(\theta) - l]_+, \quad (5)$$

$$\dot{a} = \frac{1}{\tau_a} [a_{eq}(\theta) - a - l]_+ - \dot{l}, \quad (6)$$

$$l(0) = a(0) = 0. \quad (7)$$

Here, l_{eq} and a_{eq} represent the equilibrium volume fraction of liquid and austenite phase, respectively, at temperature θ . The equilibrium volume fraction of the liquid phase $l_{eq}(\theta)$ should have the maximum value 1, when the temperature exceeds 1537°C (melting point). Tacitely neglecting a mushy zone we define $l_{eq}(\theta) = H(\theta - 1537)$, where H is the Heaviside function. In addition, we define the positive part function as $[x]_+ = xH(x)$.

For the austenite phase a , equation (6) reflects the fact that no austenite should form at the expense of liquid during heating but austenite can be transformed to liquid phase. Here, $a_{eq}(\theta)$ is the austenite equilibrium volume fraction and can be derived from a non-isothermal austenization diagram. It has value 0 below austenitization temperature A_s and then is assumed to increase linearly until the maximum value 1 is reached at A_f temperature, both depending on the chemical composition of the steel grade under consideration. There are two time constants τ_l and τ_a in equations (5) and (6) to adjust the transformation velocity. The necessary initial conditions for liquid and austenite are stated in (7).

Figure (4) shows the evolution of phase volume fractions obtained after solving (5)-(7) with a temperature history $\theta(t)$ from flame cutting. In a small fraction of time, the rapid rise in temperature takes the steel to austenization temperature and immediately into melting.

This model can be summed up as a system of rate laws (ODEs) determined by temperature θ and the already present phase fractions,

$$\dot{z}(t) = f(\theta, z), \quad (8)$$

where z is the vector of phase fractions. For the liquid and austenite model we have

$$z = (l, a)^T \quad \text{and} \quad f = (f_l(\theta, l), f_a(\theta, a, l))^T. \quad (9)$$

Due to the big size of the steel plate in comparison to the flame and the straight cutting trajectory, we now switch to a quasi-stationary state (QSS) setting to model the flame cutting heating process, allowing for more efficient simulation. The requirement for a valid QSS approach is that an observer located on the heat source, i.e., the flame, will notice no change in the temperature distribution around

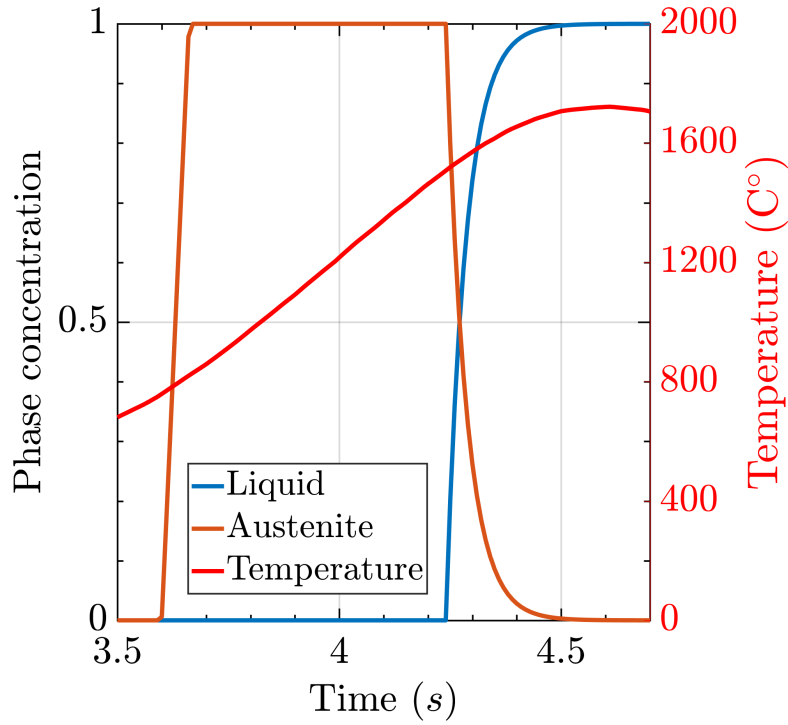


Figure 4: Evolution of volume fractions as solutions to (5)-(7) during flame cutting.

it, while the plate moves with constant velocity under the torch. To obtain the QSS equation for the phase evolution system, first we consider the equation to be defined in the whole spatial domain $\tilde{\Omega}$,

$$\dot{z}(\mathbf{x}, t) = f(\theta, z), \quad \forall \mathbf{x} \in \tilde{\Omega} \quad (10)$$

We describe the moving plate relative to the stationary heat source [11]. To this end, we replace \mathbf{x} by a new coordinate ξ subject to a given constant cutting speed \mathbf{v} , i.e., we define

$$\xi = \mathbf{x} - \mathbf{v}t \quad (11)$$

such that the phase fractions z_i are stationary, i.e., we assume

$$0 = \frac{d}{dt} z_i(\mathbf{x} - \mathbf{v}t, t) = -\mathbf{v} \cdot \nabla z_i + \frac{\partial z_i}{\partial t}.$$

Then, each component of Equation 10 can be rewritten as

$$\mathbf{v} \cdot \nabla z_i = f_i(\theta, z), \quad \forall \xi \in \Omega. \quad (12)$$

The domain Ω is stationary and has to be chosen sufficiently big.

Equation (12) is also known as transport equation. Altogether, the time-dependent evolution model of liquid and austenite phase (5)-(7) can be rewritten as the following quasi-stationary system of transport

equations,

$$\mathbf{v} \cdot \nabla l = \frac{1}{\tau_l} [l_{eq}(\theta) - l]_+, \quad \text{in } \Omega \quad (13)$$

$$\mathbf{v} \cdot \nabla a = \frac{1}{\tau_a} [a_{eq}(\theta) - a - l]_+ - \mathbf{v} \cdot \nabla l, \quad \text{in } \Omega \quad (14)$$

$$l = a = 0, \quad \text{on } \{x = 0\} \quad (15)$$

$$\frac{\partial l}{\partial \mathbf{n}} = \frac{\partial a}{\partial \mathbf{n}} = 0, \quad \text{on } \partial\Omega \setminus \{x = 0\}. \quad (16)$$

The system of equations (13)-(16) reflects the development of the liquid and austenite phase during a heating process while taking into account the movement of the domain Ω with respect to the heat source Λ . From now on, for notation purposes, we will denote the variable ξ as x and drop the t dependency. It has to be noticed that the previous initial conditions from (7) are substituted by boundary conditions. Equation (15) is a Dirichlet condition indicating that there is no volume fraction of liquid or austenite in the in-flow plane, far from the heat source. Otherwise, the homogeneous Neumann condition (16) reflects that stationarity has been achieved on the outflow boundary.

2.4 Complete System

The goal of the model is to understand the heat distribution in a moving steel plate during flame cutting with the subsequent HAZ and the trail of molten steel. The big dimensions of a steel plate (400mm long) in comparison to the heat affected zone caused by the flame (5mm) allow to consider that a quasi-stationary state (QSS) is reached in the workpiece. Accordingly, for the temperature θ in the plate, we can employ the quasi-stationary state (QSS) heat equation [11],

$$\rho(\theta) C_p(\theta) (\mathbf{v} \cdot \nabla \theta) - \nabla \cdot (\kappa(\theta) \nabla \theta) = q \quad \text{in } \Omega. \quad (17)$$

Different steel parameters are required for this equation: $\rho(\theta)$ density, $C_p(\theta)$ heat capacity and $\kappa(\theta)$ heat conductivity. One important aspect is that these properties are temperature dependent, adding non-linearities to equation (17). The term q on the right hand side is related to the heat source and therefore, to the flame and the iron burning exothermic reaction. Furthermore, this term should also include the heat absorbed and released during phase transitions, known as latent heat. For simplicity, we will restrain the model to consider only liquid and austenite phases (13)-(16), then we can write the heat source as

$$q = Q(\mathbf{v}) - \rho(\theta) L_l f_l(\theta, l) - \rho(\theta) L_a f_a(\theta, a, l). \quad (18)$$

The definition of q includes the term $Q(\mathbf{v})$ described already in subsection (2.2). The velocity \mathbf{v} is a vector which, in this case, has only x as non-zero component as the trajectory of the plate is a straight line along the x axis: $\mathbf{v} = (v, 0, 0)^T$. The heat absorbed by the phase transformations is determined by the steel density $\rho(\theta)$ and the latent heat L_i and function f_i of each phase. The functions are specified in equations (13) and (14).

Following subsection (2.3), we add the QSS phase equations for austenite and liquid as they are coupled with the heat equation due to being temperature dependant. These equations with adequate boundary conditions comprise the system of equations (19)-(29). Considering θ_a as the ambient temperature, Newton's cooling law is imposed on the top and bottom of the plate (22)-(23) to describe

the exchange of heat with surrounding air with the convection factors h_1 and h_2 . The vector \mathbf{n} is an outward normal unit vector to the corresponding surface. The temperature of the plate before the process is θ_a and far enough from the torch, the temperature suffers no change. Therefore, a Dirichlet condition is enforced on the plane $\{y = y_{max}\}$ and on the inflow plane $\{x = 0\}$, see equations (24)-(26). Finally, on the symmetry plane $\{y = 0\}$ and on the outflow boundary $\{x = x_{max}\}$, a homogeneous Neumann condition is imposed implicating that there is no heat flux in the normal direction. The boundary conditions for the phase fractions have already been explained above.

$$\rho(\theta) C_p(\theta) (\mathbf{v} \cdot \nabla \theta) - \nabla \cdot (\kappa(\theta) \nabla \theta) = q, \quad \text{in } \Omega \quad (19)$$

$$\mathbf{v} \cdot \nabla l = f_l(\theta, l), \quad \text{in } \Omega \quad (20)$$

$$\mathbf{v} \cdot \nabla a = f_a(\theta, a, l), \quad \text{in } \Omega \quad (21)$$

$$-\kappa(\theta) \frac{\partial \theta}{\partial \mathbf{n}} = h_1(\theta - \theta_a), \quad \text{on } \{z = 0\} \quad (22)$$

$$-\kappa(\theta) \frac{\partial \theta}{\partial \mathbf{n}} = h_2(\theta - \theta_a), \quad \text{on } \{z = z_{max}\} \quad (23)$$

$$\theta = \theta_a, \quad \text{on } \{x = 0\} \quad (24)$$

$$\frac{\partial \theta}{\partial \mathbf{n}} = 0, \quad \text{on } \{x = x_{max}\} \quad (25)$$

$$\theta = \theta_a, \quad \text{on } \{y = y_{max}\} \quad (26)$$

$$\frac{\partial \theta}{\partial \mathbf{n}} = 0, \quad \text{on } \{y = 0\} \quad (27)$$

$$l = a = 0, \quad \text{on } \{x = 0\} \quad (28)$$

$$\frac{\partial l}{\partial \mathbf{n}} = \frac{\partial a}{\partial \mathbf{n}} = 0, \quad \text{on } \partial\Omega \setminus \{x = 0\}. \quad (29)$$

3 Numerical Approach

The purpose of the model presented is to predict the temperature distribution, the HAZ size in terms of the fraction of austenite and the kerf, i.e., the trail of molten steel produced by flame cutting of a high strength steel plate with 40 mm thickness. We apply the Finite Elements Method (FEM) to solve the system (19)-(29) and the requirements for this are detailed in the following.

The material properties required for the model are the density, specific heat and thermal conductivity of the considered Raex[®] 400 steel grade. These temperature dependant properties were obtained using the commercial software JMatPro[®] [12], which mainly requires as input the chemical composition of the steel (Table 1).

| C | Si | Mn | P | S | Cr | Ni | Mo | B |
|---------|---------|---------|---------|---------|---------|---------|---------|---------|
| (max %) | (max %) | (max %) | (max %) | (max %) | (max %) | (max %) | (max %) | (max %) |
| 0.23 | 0.80 | 1.70 | 0.025 | 0.015 | 1.50 | 1 | 0.50 | 0.005 |

Table 1: Chemical composition of Raex[®] 400. The steel is grain refined.

After defining the geometric parameters and constants of the model, it is possible to solve the system of equations (19)-(29). This is achieved using the finite element package *pdelib2* developed and

maintained at WIAS. The non-linearities together with the coupling were solved using a fixed-point algorithm and an adaptive mesh module was used to refine the mesh based on a residual error estimator resolving regions with steep gradients for the temperature and phase fractions [13].

Note that the phase transition model (20)-(21) represents a system of transport equations and cannot be solved directly using FEM in 3D without encountering erroneous results [14]. The first step taken to deal with this problem is to add artificial diffusivity to the equation. This consists in adding a diffusion term with a coefficient $\varepsilon > 0$.

$$\mathbf{v} \cdot \nabla z - \varepsilon \Delta z = f(\theta, z) \quad (30)$$

The value of ε must be small enough to avoid too much diffusion but big enough to be effective. For our simulations, we found $\varepsilon = 10^{-8}$ to produce an adequate outcome, but still a further step is required. Equation (30) is a steady-state linear scalar convection-diffusion equation with a convection-dominated regime. It is well known that, with this regime, the solutions from FEM contain spurious oscillations. A stabilization method is required. We chose the Streamline-Upwind Petrov-Galerkin (SUPG) method that adds diffusion in the direction of the streamlines [14].

In order to highlight the need for the aforementioned steps to solve appropriately equations (20)-(21) when using FEM in a 3D domain, Figure 5 represent the numerical results for the liquid phase in each mentioned step. A temperature distribution in a plate was retrieved after solving the heat equation with a flame cutting. This temperature field was used as input to solve directly the transport equation (20) leading to erratic results (see figure on the left). Then, the result obtained with the addition of artificial diffusivity (30) is in the middle of Figure 5 while the figure on the right displays the final result with the introduction of the SUPG method.

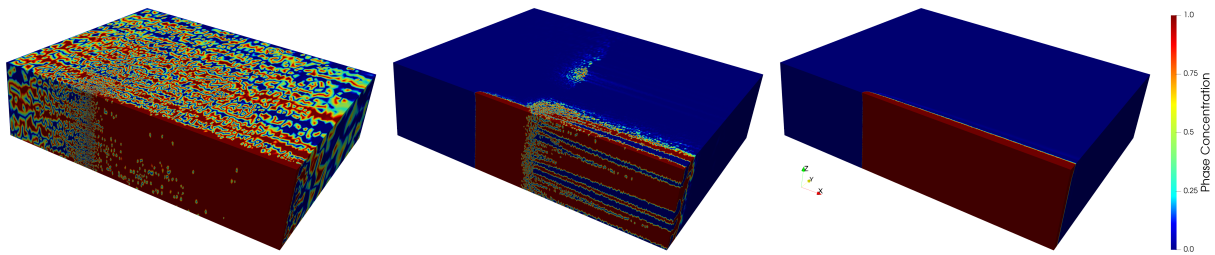


Figure 5: Numerical solution of the liquid phase equation with FEM. Left: full convection equation (13), middle: with artificial diffusivity (30), right: with artificial diffusivity and SUPG stabilization.

4 Results and Validation

Experimental data from two experiments are used to calibrate the volumetric heat source. Both experiments consider a 40 mm Raex[®] 400 steel plate but different cutting speed: 135 and 270 mm/min. The results comprise isothermal curves in different planes transversal to the cutting direction, see Figure 2. In this section, the main results presented belong to the 135 mm/min case as it was used as reference case. Furthermore, the results obtained for 135, 202.5 and 270 mm/min cutting speeds are compared.

From the experimental data, the shape of the 1537 °C isothermal line in Figure 2 plus the estimated size of the kerf of 1 mm for the case with a cutting speed of 135 mm/min suggested to use the same values for the volumetric heat source as Thièbaud [1]. These values are $r_1 = 2$ mm, $r_2 = r_3 = 1$ mm and $z_1 = z_2 = 2$ mm. The power within Λ was set to 7.35 kW. The size of the computational domain was established to be in accordance with the Dirichlet conditions: $y_{max} = 100$ mm and $x_{max} = 140$

mm. The vertical axis of the heat source Λ was located at (40 mm, 0). Heat transfer coefficients h_1 and h_2 were set to 5 W/(m²K) and the external temperature θ_a was 25 °C. Regarding the phase equations, the latent heat values were 272 and 16 kJ/kg for liquid and austenite, respectively. The values of τ_l and τ_a representing the transformation velocity were 1E-2 and 5E-1. It was found that is important that τ_l is smaller than τ_a to produce a small interphase region.

4.1 Case 135 mm/min

The main focus was on this case as it was used as the reference case. A general view of the temperature distribution around the heat source is given in Figure 6. In this figure, the features of Λ can be sensed from the isothermal surfaces slightly above 1500 °C. The maximum temperature of 3000 °C is reached on the top of the plate in the torch centre, and the temperature is dissipated according to the direction of the movement. The corresponding trail of liquid phase (molten steel) and austenite phase produced with the heat source is shown in Figure 7.

Both phases start approximately at the isothermal surface corresponding to the temperature where the equilibrium volume fraction of each phase is 1, then the phases are transported in the direction of the movement, the x -direction. The trail of liquid is the closest to the cutting plane as expected from the higher temperatures being in this plane. In Figure 7, there is a transparent slice of the liquid and austenite volumes such that the inner profile of both phases along the thickness of the plate can be seen. The liquid phase is wider on top while in the middle and bottom it remains the same. On the other hand, there is no difference in austenite width along thickness.

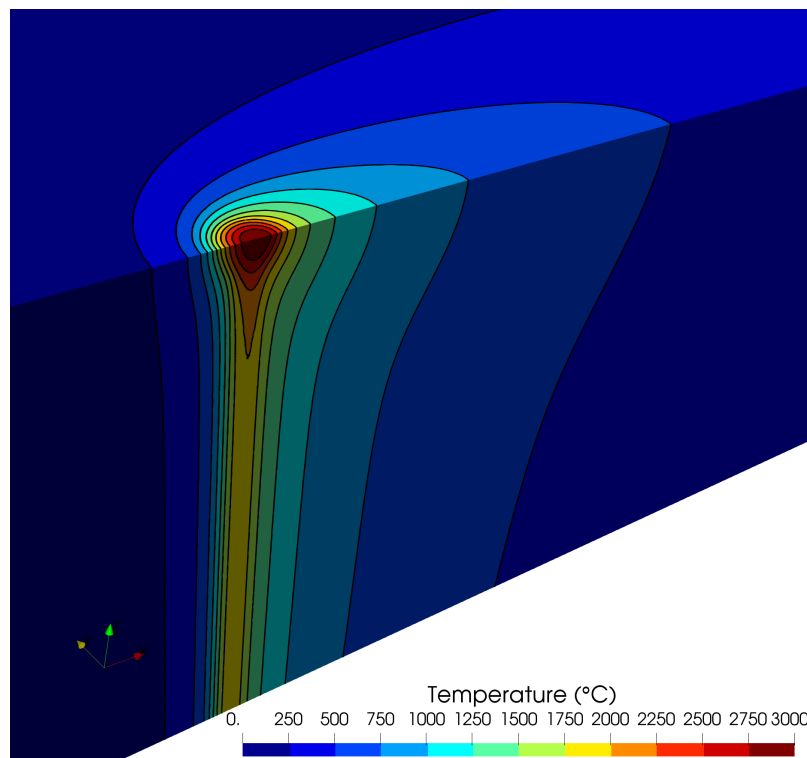


Figure 6: Temperature distribution around the heat source location. Case 135 mm/min.

To understand more clearly the next results, Figure 8 illustrates the location of the straight lines A, B, C and D along which the results were extracted. In Figure 9, we can see the phases evolution and temperature along lines A (left) and B (right). As A crosses a zone of rapid increase in temperature

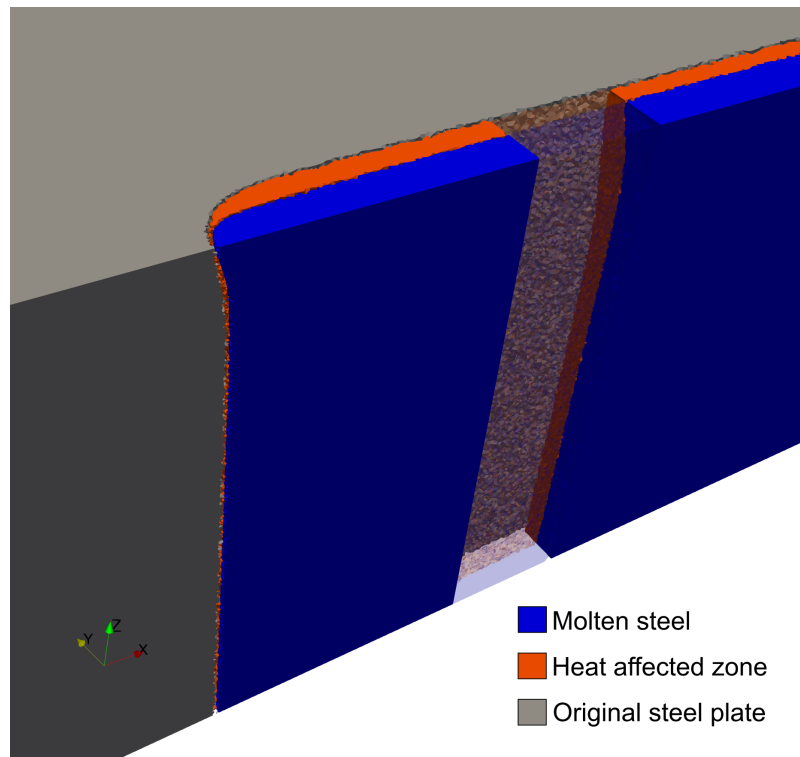


Figure 7: Trail of liquid phase (blue) and austenite (orange) after flame cutting. Case 135 mm/min.

the austenite phase grows very quickly until the melting point is reached and it decreases while the liquid phase grows. On the other hand, the B line is far enough from the torch and the temperature only causes austenite growth.

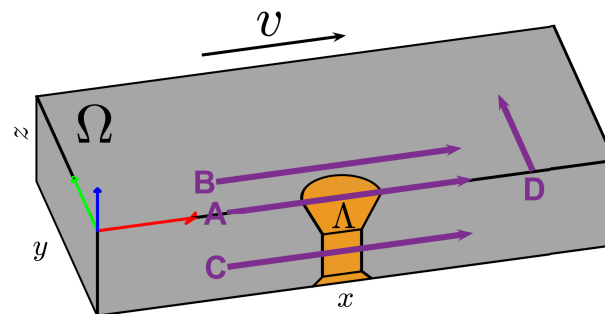


Figure 8: Domain with purple lines where results were analysed.

As the phases can be identified with the kerf and the Heat Affected Zone (HAZ), we can retrieve the width of both areas with line D. This is presented in Figure 10. On the cutting plane we find the liquid phase, then a very small two-phase region and finally the remaining plate with an austenized and the non-affected area. Furthermore, as D is in the transversal plane to the cutting, we can see how the temperature dissipates in the y -direction. The temperature values observed in the immediate area after the austenite region indicate that tempering of the steel occurs [7].

A rapid increase in temperature in the plate and the cooling afterwards can be observed in Figure 11. We track the temperature on the top (left) and middle (right) of the plate using lines A and C. Moreover, these lines are shifted in the y -direction to obtain more information about temperatures reached in the remaining plate. As shown in Figure 7, the maximum temperature is reached on the top of the plate. It

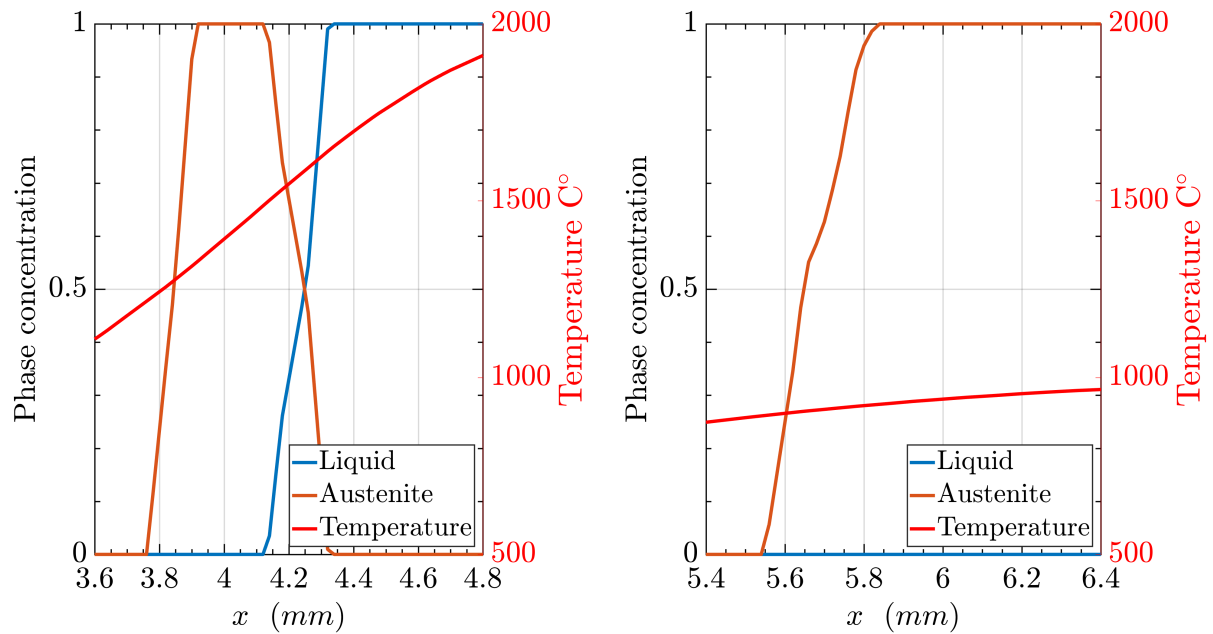


Figure 9: Evolution of phases and temperature along lines A (left) and B (right).

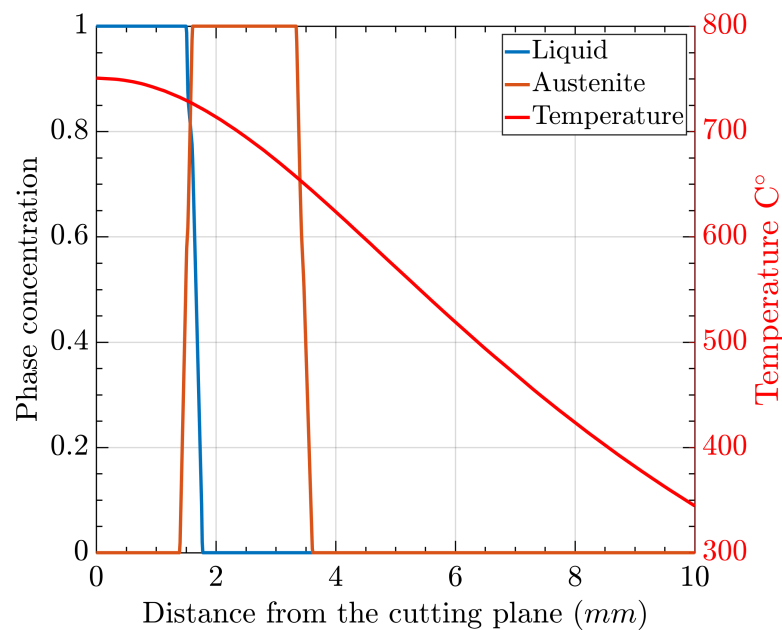


Figure 10: Liquid and austenite phases and temperature along line D

is interesting to notice that the maximum is not reached in the axis of Λ , $x = 40$ but slightly later due to the movement.

4.2 Case 202,5 and 270 mm/min

To see the effect of different cutting speeds we multiplied the base speed 135 mm/min, by a factor 1.5 and 2, resulting in 202.5 and 270 mm/min. As it is explained in the model, the velocity v affects the

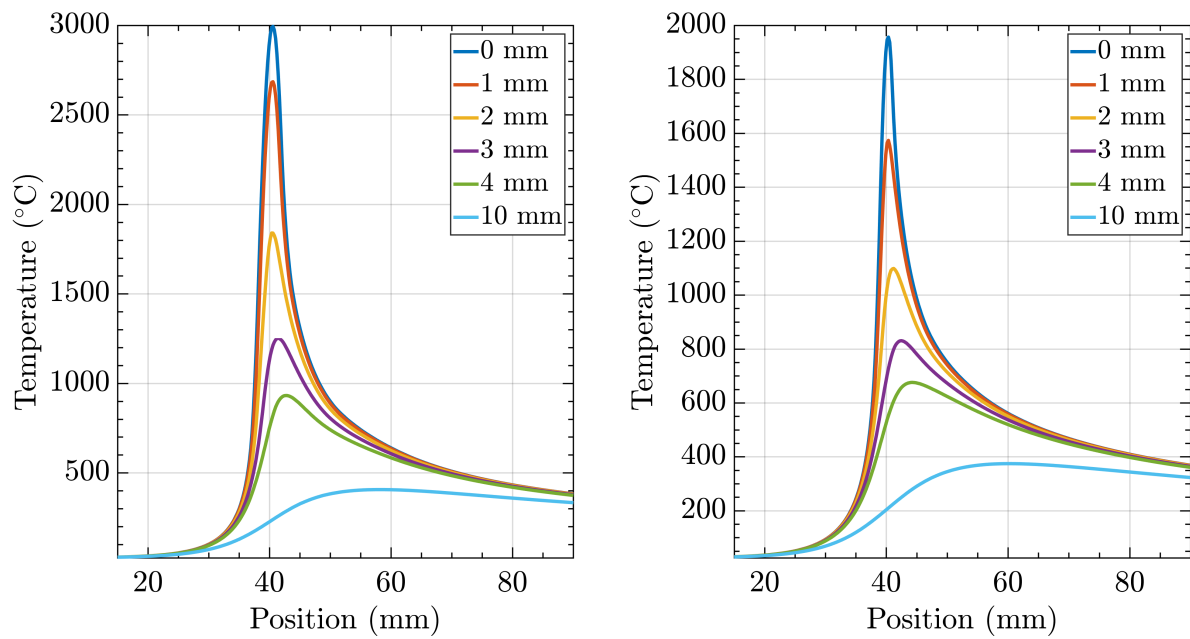


Figure 11: Evolution of nodal temperature from numerical simulation along lines A (left) and C (right). Lines A and C were shifted to distances of 1, 2, 3, 4, and 10 mm from the cutting plane.

volume Λ where the heat source is defined. Specifically the speed affects the value of $C(v)$ which scales the radii of Λ with speed (cf. (2)). The function $C(v)$ is assumed to be linear and monotonically decreasing. More precisely, for the base case, we choose $C(135) = 1$, while $C(202.5) = 0.75$ and $C(270) = 0.5$.

The features of the results obtained with faster velocities are the same qualitatively. Quantitatively, with higher speed, the influence area of the torch and the temperature distribution produce a smaller kerf and HAZ. In Table 2, the average values of kerf and HAZ width for the three cases are shown. Table 2 provides a good insight of the trends: smaller region width with increasing speed and greater values on the top than in the middle and at the bottom of the plate.

| | Kerf width (mm) | | HAZ width (mm) | |
|--------------|-----------------|-------------------|----------------|-------------------|
| | Top | Middle and bottom | Top | Middle and bottom |
| 135 mm/min | 2.15 | 1.00 | 1.57 | 1.58 |
| 202.5 mm/min | 1.68 | 0.67 | 0.96 | 0.93 |
| 270 mm/min | 1.33 | 0.4 | 0.57 | 0.41 |

Table 2: Kerf and Heat Affected Zone (HAZ) width after flame cutting on the top, middle and bottom of the plate for different velocities.

As the kerf width is known after the numerical simulation, we can focus on the remaining plate. In Figure 12, the temperature fields for the three cases are aligned to the resulting cutting edge. Thus, we can observe the difference in the temperature reached within the plate. As expected, with slower speed higher temperatures reach further into the plate.

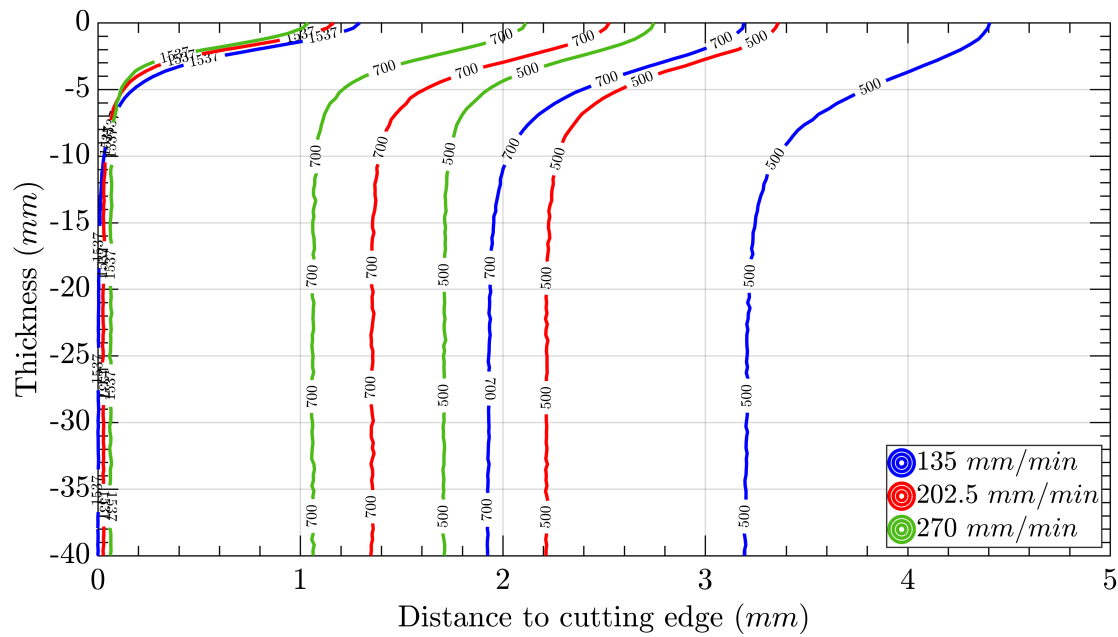


Figure 12: Isothermal lines in the remaining plate for different velocities.

| | Fuel | | °C | K |
|--|-----------|----------|------|------|
| | Acetylene | C_2H_2 | 3069 | 3342 |
| | Ethylene | C_2H_4 | 2902 | 3175 |
| | Propane | C_3H_8 | 2822 | 3095 |

Table 3: Adiabatic flame temperature for fuels combusted with oxygen [15].

5 Discussion of results and conclusions

The model developed provides relevant information about the temperature distribution along the steel plate during the flame cutting process. The validated results can be used as a temperature history in every coordinate of the plate, allowing further studies involving solid phase changes or residual stresses. In general there is a good correspondence between the temperature values from experimental experience and those obtained after numerical simulation.

Two different approaches are possible for consideration of removal of material during the simulation [4, 5], either the domain starts in the cutting plane (torch position) or in the cutting edge (resultant edge after cutting). In Thiébaud [1] a mix of both options is used as the domain includes a predefined kerf for the area already affected by the flame. In this work, the domain starts in the cutting plane allowing us to describe the kerf size as the liquid phase created during flame cutting instead of predefining it.

The heat source Λ is defined to include heat from the burning of fuel gas and the exothermic reaction of burning of iron. The maximum temperature values obtained from the simulations are in a realistic range. These values can be drawn from Figure 11. The maximum temperature on top is 3000°C and 2000°C in the distance of half the thickness from the top surface of the plate. The maximum temperatures of the flame in adiabatic conditions for common fuel gases are displayed in Table 3 [15]. The values can only be used as reference since losses are not considered nor the heating by the oxidation reaction. While in Jokiaho [4] temperatures slightly over 1600°C are reached in the cutting edge, Thiébaud [1] using the data for propane arrives at about 2500°C in the cutting line. Temperature gradients with respect to plate depths are hardly mentioned in literature.

The disagreement between numerical results and experimental data is higher in the surface area (compare Figs. 2 and 12). More heat should be spread in the y -direction. In practice, this effect is caused by a deflection of the flame on top of the plate. A more complex definition of the heat source in Bae et. al. [5] or [8] may provide more realistic results. For example in [5], it is divided into three terms, one of them is a Gaussian function which is only defined in the upper surface of the plate and may help balancing the temperatures obtained numerically, producing less heat in the cutting line but more into the width of the plate as should be expected according to experimental data.

Thiébaut and Jokiahho [1, 4] get similar temperature profiles to the ones displayed in Figure 11. The total power required by Thiébaut is 18.8 kW, which is higher than the one estimated in this study, 7.35 kW. This difference is mostly related to the different treatment of the kerf as explained before.

The inclusion of quasi-stationary phases transition models for liquid and austenite in the present paper provides a new tool for a deeper understanding of the flame cutting process. The liquid phase fraction allows for a rather precise description of the kerf while the austenite fraction does the same for the heat affected zone. The trend found of decreasing width of the HAZ with increasing speed (Table 2) is also observed in experimental results in Vippola et al. [7]. In their experiments the HAZ is assumed to be formed by a martensitic region, a two phase region and a tempered region.

It is fair to conclude that the model derived in the present paper covers all essential features of the flame cutting process of steel. For more reliable quantitative predictions a more precise adjustment of phenomenological model parameters based on refined measurements will be necessary. In a forthcoming paper, the optimal control of the flame cutting process in order to mitigate the danger of cold cracks after the cutting process will be discussed [16].

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